Seat No.:	Enrolment No.
Seat No	Linomient No.

GUJARAT TECHNOLOGICAL UNIVERSITY

BE - SEMESTER-VII • EXAMINATION - WINTER • 2014

	•	Code: 170104 Date: 27-11-2014	
Tim	_	Name: Rocket and Missile Configurations Design 0:30 am - 01:00 pm Total Marks: 70	
mstr		Attempt all questions. Make suitable assumptions wherever necessary.	
Q.1	(a)	Define ullage? Explain pressurization system with neat sketch.	07
	(b)	Explain the effects of Aspect ratio on the wing of missile.	07
Q.2	(a)	What are the major considerations of good load controls in propellant loading tolerance? Explain in brief.	07
	(b)	Write a short note on elimination of the Geysering effect in missiles.	07
	(b)	OR Explain pressure surge in propellant system feed line flows.	07
Q.3	(a)	Explain solid propellant rocket engine with neat sketch.	07
	(b)	Explain liquid propellant properties in details.	07
Q.3	(a)	OR Explain liquid propellant rocket engine with neat sketch.	07
	(b)	Write a short note on desirable physical properties for liquid propellant.	07
Q.4	(a)	Explain advantages and disadvantages of wing control, Canard control and Tail control.	07
	(b)	Write a short note supersonic wing planforms.	07
Q.4	(a)	OR What is outage? Explain calibrated system outage control.	07
	(b)	What are the basic problems must be considered in propellant tank outlet design? Explain in details.	07
Q.5	(a)	Explain with neat sketch Long range cruise trajectory.	07
	(b)	What is cruciform? Explain with neat sketch.	07
Q.5	(a)	OR Relation of maneuverability and static stability margin for maneuvering flight.	07
	(b)	Explain with neat sketch Boost sustain trajectory.	07
